

18AE/AS42

## Fourth Semester B.E. Degree Examination, June/July 2023 Aerodynamics – I

Time: 3 hrs.

Max. Marks: 100

Note: Answer any FIVE full questions, choosing ONE full question from each module.

Module-1

- a. Derive the integral and differential form of continuity equation by control volume approach.

  (10 Marks)
  - b. Define the following:
    - (i) Streak line
    - (ii) Path line
    - (iii) Circulation
    - (iv) Velocity potential function.
    - (v) Stream function

(10 Marks)

OR

- 2 a. Derive the integral and differential form of momentum equation by control volume approach. (10 Marks)
  - b. Derive the relationship between:
    - (i) Vorticity and circulation
    - (ii) Stream function and Velocity potential function

(10 Marks)

Module-2

- 3 a. Derive the relation to calculate the Aerodynamic forces N' and A' and the momentum  $M'_{LE}$  in terms of P,  $\theta$  and  $\tau$ . (10 Marks)
  - b. With neat sketches, explain (i) Airfoil nomenclature (ii) Wing planform geometry.

(10 Marks)

OR

4 a. In detail explain different types of Drag

(10 Marks)

- b. Explain the following:
  - (i) Aerodynamic centre
  - (ii) Centre of pressure
  - (iii) Pressure coefficient
  - (iv) Aerodynamic forces and moments

(10 Marks)

Module-3

5 a. Derive an expression non lifting flow over a circular cylinder.

(10 Marks)

- b. Define: (i) Source flow
- (ii) Sink flow
- (iii) Doublet

- (iv) Vortex flow
- (v) Uniform flow

(10 Marks)

(10 Marks)

OR

Derive an expression for left co-efficient for symmetrical airfoil, using classical thin airfoil (10 Marks) theory. b. Briefly explain the following, with neat sketches and relevant expressions: Kelvin's circulation theorem. (i) (10 Marks) The starting vortex. (ii) Module-4 Discuss the following: Biot-Savart law and Helmholtz's theorems. (i) Downwash and induced drag. (10 Marks) (ii) (10 Marks) Derive Prandtl's classical lifting line theory. OR Prove that induced drag co-efficient is directly proportional to square of left co-efficient (10 Marks) using elliptical lift distribution. b. Obtain the fundamental equation of Prandtl's lifting-line theory. (10 Marks) Module-5 (10 Marks) Describe simplified horse-shoe vortex model. Explain the following: Formation flight. (i) (10 Marks) Ground effects. (ii) Write a short note on: 10 Drag divergence mach number. (i) Subsonic and supersonic leading edges. (10 Marks) (ii) Write a short notes on:

Trans sonic area rule. Critical Mach number.

(ii)