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18AE71

# Seventh Semester B.E. Degree Examination, July/August 2022 **Aircraft Stability and Control**

Time: 3 hrs.

Max. Marks: 100

Note: Answer any FIVE full questions, choosing ONE full question from each module.

## Module-1

Explain Longitudinal Static Stability.

(06 Marks)

- Using a neat sketch, describe the forces and moments acting on an Airplane.
- (04 Marks) c. Derive an expression for Tail contribution  $\left(\frac{C_{\text{M}}}{C_{\text{L}}}\right)$  for the static longitudinal stability of an

Airplane tail and discuss the downwash at the tail.

(10 Marks)

## OR

Define Stick Fixed Neutral point and Static Margin.

(04 Marks)

- b. Derive an expression for Stick Fixed Neutral point and discuss the effect of CG shift on pitching moment. (10 Marks)
- Derive the expression for Elevator control power:  $C_{m\delta_a} = -V_H \eta C_{L_{at}} \tau$ .

(06 Marks)

## Module-2

- Briefly discuss Elevator hinge moment parameters and trim tab with suitable sketches and 3 equations. (10 Marks)
  - b. Obtain the expression for neutral point for stick free condition with necessary graphs.

(10 Marks)

#### OR

Write short notes on Aerodynamic balancing and its types.

(10 Marks)

Explain Stick Force Gradient conditions in unaccelerated flight in detail.

(10 Marks)

#### Module-3

- a. Briefly explain the requirements of directional control and obtain the expression for rudder control effectiveness  $C_{n\delta_r}$ . (10 Marks)
  - b. Enumerate about "Rudder Lock" and "Dorsal Fin".

(06 Marks)

c. Discuss about Static directional stability for Stick – Free conditions.

(04 Marks)

#### OR

Explain Dihedral Effect.

(06 Marks)

Discuss the effect of Wing sweep, Flaps and power on dihedral effect, with neat sketches. (10 Marks)

What is meant by "Adverse Yaw Effect" and "Aileron Reversal" in Static Lateral Stability? (04 Marks)

### Module-4

Derive Rigid Body equation of Motion. 7

(12 Marks)

Briefly explain Gravitational and Thrust forces acting on the Airplane.

(08 Marks)

OR

8 a. Starting with X – Force equation, use the Small Disturbance theory to determine the linearised force equation. Assume a steady level flight for the reference flight conditions.

(10 Marks)

b. Obtain derivatives due to the time rate of change in angle of attack.

(10 Marks)

Module-5

9 a. Explain Routh's criteria and determine whether the characteristic equations given below have stable or unstable roots:

i)  $\lambda^3 + 6\lambda^2 + 12\lambda + 8 = 0$ 

(i)  $2\lambda^3 + 4\lambda^2 + 4\lambda + 12 = 0$ .

(10 Marks)

b. Discuss the following:

- i) Cooper Harper Scale
- ii) Effect of wind shear.

(10 Marks)

OR

10 a. Explain the dynamic lateral stability considering rudder Free case with expressions.

(10 Marks)

b. Discuss the following:

- i) Dutch roll and Spiral instability.
- ii) Auto rotation and Spin.

(10 Marks)