USN

18AE/AS35

Third Semester B.E. Degree Examination, July/August 2022 Mechanics of Fluids

Time: 3 hrs.

Max. Marks: 100

Note: Answer any FIVE full questions, choosing ONE full question from each module.

Module-1

a. State Newton's law of viscosity and explain about types of fluid based on Newton's law.
(06 Marks)

b. Give reasons for the following:

i) Rain drops, water droplets and Bubbles are spherical in shape.

ii) Viscosity of liquids decreases with increase in temperature by viscosity of gas increases with increase in temperature. (06 Marks)

c. If the velocity profile of a fluid over a plate is parabolic C with the vertex 20cm from the plate, where the viscosity is 120cm/sec. Calculate the velocity gradient and shear stresses at a distance of 0cm, 10cm and 20cm from the plate. Assume $\mu = 8.5$ poise. (08 Marks)

OR

- 2 a. Prove that pressure intensity at a point in static fluid is same in all directions. (06 Marks)
 - b. Write different types of manometers and explain any two types with neat sketch. (06 Marks)
 - c. A triangular plate of base 2m and height 3m which is immersed in water in such a way that plan of plate makes an angle of 60° with the free surface of water. Base of plate is parallel to water surface and at a depth of 2.5m from water surface. Find total pressure and position of centre of pressure.

 (08 Marks)

Module-2

a. Explain different types of fluid flow.

(06 Marks)

- b. Water flows through a pipe AB of 1.2m in diameter at 3m/s and then passes through a pipe BC of 1.5m diameter. At point C, the pipe branches as CD and CE where CD is 0.8m in diameter and carries one third of the flow in AB. Flow velocity in CE is 2.5m/s. Find the volume rate of flow in AB, velocity in BC, velocity in CD and diameter of CE. (08 Marks)
- c. Draw and explain about: i) Source ii) Sink iii) Doublet. (06 Marks)

OR

- a. Define velocity potential and stream function and write its expression for velocity components prove that the product of equipotential lines and stream line is (-1). (08 Marks)
 - b. Derive energy equation in integral form and explain its application. (12 Marks)

Module-3

- 5 a. Derive Euler's equation and using that derive Bernoulli's equation. Also write the assumptions made for Bernoulli's equation. (08 Marks)
 - b. A vertical venturimeter has an area ratio 5. If has a throat diameter of 10cm. When an oil of specific gravity 0.8 flows through it, the mercury differential manometer in the differential gauge indicates a height difference of 12cm. Find the discharge through venturimeter if co-efficient of discharge is 0.98.
 (06 Marks)
 - c. Explain about pitot-tube and write about different types of arrangements needed for pitot-tube to measure velocity. (06 Marks)

2. Any revealing of identification, appeal to evaluator and /or equations written eg, 42+8 = 50, will be treated as malpractice. Important Note: 1. On completing your answers, compulsorily draw diagonal cross lines on the remaining blank pages.

OR

- a. Find the expression for ΔP in a pipe which depends on diameter D, length ' ℓ ', velocity V, Viscosity μ , density ρ and roughness 'K', using Buckingham's π - theorem.
 - b. The pressure drop in an aircraft model of size 1/50 of its prototype is 4N/cm². The model is tested in water. Find the corresponding pressure drop in prototype. Take $\rho_{air}=1.24kg/m^3$, (06 Marks) $\mu_{\text{water}} = 0.01 \text{ Poise}, \ \mu_{\text{air}} = 0.0018 \text{ Poise}.$ (04 Marks)
 - Write about types of forces acting in a moving fluid.

Module-4

- Derive the relation for momentum thickness and energy thickness for boundary layer. (08 Marks) 7
 - b. A kite weighting 7.848N has an area of 0.8m². It is flying in air at an angle of 10° to the horizontal. The string attached to the kite makes an angle of 45° to the horizontal and at this position the value of $C_D = 0.6$ and $C_L = 0.8$. Find the speed of the wind and tension in the (04 Marks)
 - State and explain Kutta Joukowsky theorem.

OR

- Derive Von-Karman momentum Integral equation.
 - Find the displacement thickness, momentum thickness and energy thickness for the velocity distribution in the boundary layer given by $\frac{u}{U} = \frac{y}{\delta}$ (10 Marks)

Module-5

- Derive the expression for velocity of sound wave and velocity of sound for adiabatic (08 Marks)
 - b. Find the Mach number when an aircraft is flying at 1100km/hr through still air having a pressure of 7N/cm² and temperature of -5°C. Calculate the pressure, temperature and density of air at stagnation point on the nose of the point aircraft. Take K = 1.4, R = 287.14 J/kg-K.
 - c. Draw a normal shock wave and express the property changes before and after the shock.

OR

- Draw the propagation of pressure waves for different mach numbers and explain about Mach wave, Mach cone and Mach angle.
 - b. For an adiabatic flow, A gas with a velocity of 300m/s is flowing through a horizontal pipe at a section where pressure is $6 \times 10^4 \text{N/m}^2$ and temperature 40°C. The pipe changes in diameter and at this section pressure is $9 \times 10^4 \text{N/m}^2$. Find the velocity of gas at this section. (10 Marks) Take K = 1.4 and R = 287 J/kg-K.