

OR

- 6 a. Explain hydrodynamic boundary layer theory and thermal boundary layer theory with suitable figures. (08 Marks)
- b. Air flows through a long rectangular (30cm height \times 60cm width) air conditioning duct maintains the outer duct surface temperature at 15°C. If the duct is uninsulated and exposed to air at 25°C, calculate the heat gained by the duct per meter length, assuming it to be horizontal. (12 Marks)

Module-4

- 7 a. Explain following :
 i) Specular reflection and diffuse reflection
 ii) Kirchoff's law
 iii) Lamberr's cosine law
 iv) Black body. (08 Marks)
- b. Emissivities of two large parallel plates maintained at 800°C and 300°C are 0.3 and 0.5 respectively. Find the net radiant heat exchange per square meter for these plates. (04 Marks)
- c. The net radiation from the surfaces of two parallel plates maintained at temperature T_1 and T_2 is to be reduced by 79 times. Calculate the number of screens to be placed between the two surfaces to achieve this reduction in heat exchange, assuming the emissivity of the screen as 0.05 and that of the surfaces as 0.8. (08 Marks)

OR

- 8 a. Obtain expression for LMTD of counter flow heat exchanger. (08 Marks)
- b. Water enters a cross flow heat exchanger (both fluids unmixed) at 5°C and flows at the rate of 4600kg/n to cool 4000kg/n of air that is initially at 40°C. Assume overall heat transfer coefficient to be 150w/m²K and area of 25m², calculate the exit temperature of air and water. Take $C_{pw} = 4.18$ kJ/kg k and $C_{pair} = 1.01$ kJ/kgk. (08 Marks)
- c. A counter flow concentric flow heat exchanger is used to cool the engine oil [$C_p = 2130$ J/kgk] from 160°C to 60°C with water available at 25°C as the cooling medium. The flow rate at the cooling water of inner diameter of 0.5m is 2 kg/s, while flow rate of oil through outer annulus of diameter 0.7m is also 2kg/s. IF $U = 300$ w/m²k, How long must be the heat exchanger to meet cooling requirement. (04 Marks)

Module-5

- 9 a. Explain heat distribution in rocket thrust chamber. (08 Marks)
- b. Explain ablative heat transfer. (08 Marks)
- c. Explain aerodynamic heating in Aerospace engineering. (04 Marks)

OR

- 10 a. Obtain species conservation equation using conventional notations. (10 Marks)
- b. Ambient air at 20°C, flows past a flat plate with a sharp leading edge at 3m/sec. The plate is heated uniformly throughout its entire length and it is maintained at a surface temperature of 40°C. Calculate the distance from leading edge at which the flow in the boundary layer changes from laminar to turbulent. Assume transition occurs at a critical Reynolds number of 5×10^5 . Determine :
 i) Thickness of hydrodynamic and thermal boundary layer at transition point
 ii) Local and average heat transfer coefficient
 iii) Total drag per unit width on one side of plate
 iv) Convective heat flow from plate to ambient air considering unit width of the plate. (10 Marks)
