## USN

## Fifth Semester B.E. Degree Examination, Dec.2018/Jan.2019 Aircraft Propulsion

Time: 3 hrs.

Max. Marks:100

Note: Answer any FIVE full questions, selecting at least TWO full questions from each part.

## PART - A

- 1 a. What is meant by jet propulsion? What are the classifications of power plants used in aircrafts? (06 Marks)
  - b. Define stagnation enthalpy, stagnation temperature, stagnation pressure, stagnation density.
  - c. Explain Fourier's law of heat conduction, Newton law of cooling and Stefan Boltzmann law.
    (06 Marks)
- 2 a. With a neat sketch, explain the working principle of Turbofan engine. Indicate the advantages and disadvantages of Turboprop engine. (08 Marks)
  - b. What are the various factors affecting thrust based on operating characteristics of the engine?
  - c. The effective jet exit velocity from a jet engine is 2700 m/s. The forward flight velocity is 1350 m/s and the air flow rate is 78.6 kg/s calculate thrust, thrust power and propulsive efficiency.

    (06 Marks)
- 3 a. Derive a relation for minimum area ratio  $\left(\frac{A_{max}}{A_i}\right)$  min in terms of external deceleration ratio
  - $\frac{u_i}{u}$  and explain with relevant graphs.

(10 Marks)

- b. Explain the following with suitable sketches: i) Vortex generators ii) Supersonic inlets.
  (10 Marks)
- 4 a. What are the various factors affecting combustor design? (06 Marks)
  - b. Explain different types of combustion chambers used in gas turbine engines with suitable sketches.

    (06 Marks)
  - c. Write shot notes on: i) Thrust vectoring ii) Thrust reversing.

(08 Marks)

## PART - B

- 5 a. With a neat sketch, explain the essential parts of a centrifugal compressor. (06 Marks)
  - b. Define degree of reaction of axial flow compressor and derive an expression for 50% degree of reaction.

    (06 Marks)
  - c. A centrifugal compressor compresses 30kg of air per second at a rotational speed of 15000 rpm. The air enters the compressor axially and the conditions at the exit sections are radius = 0.3m, relative velocity of air at the tip = 100 m/s at an angle of 80°C with respect to plane of rotation. Take  $P_{01} = 1$  bar and  $T_{01} = 300$  K. Find the torque and power required to drive the compressor, work done and head developed. (08 Marks)

6 a. Explain with a neat sketch the principle of operation of single stage reaction turbine.

(06 Marks)

- b. Write a neat sketch of velocity triangle of single stage axial turbine and derive its expression. (06 Marks)
- c. Discuss the various types of losses in a turbine.

(08 Marks)

- 7 a. Discuss the operation of Ramjet engines mention its advantages and disadvantages characteristics. (10 Marks)
  - b. Explain critical, subcritical and supercritical operation of combustion in ramjet engine.

(10 Marks)

- 8 a. With a help of a schematic diagram explain the principle of operation of liquid propellant rocket. Indicate its advantages over solid propellant rockets. (10 Marks)
  - b. Explain ion rocket propulsion with a relevant sketch.

(06 Marks)

c. Write a short note on multistage rocket.

(04 Marks)